





### D-A183 555

AD

TECHNICAL REPORT ARCCB-TR-87018

OTTE FILE COPY

# ANALYSIS OF COMPOSITE SHRINK FITS TRESCA MATERIAL

PETER C. T CHEN



**JULY 1987** 



## US ARMY ARMAMENT RESEARCH, DEVELOPMENT AND ENGINEERING CENTER

CLOSE COMBAT ARMAMENTS CENTER BENÉT WEAPONS LABORATORY WATERVLIET, N.Y. 12189-4050

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

ECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)	A183 55
REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
	N NO. 3. RECIPIENT'S CATALOG NUMBER
ARCCB-TR-87018	
. TITLE (and Subsisse) ANALYSIS OF COMPOSITE SHRINK FITS —	S. TYPE OF REPORT & PERIOD COVERED
TRESCA MATERIAL	Final
	6. PERFORMING ORG. REPORT NUMBER
. AUTHOR(e)	8. CONTRACT OR GRANT NUMBER(*)
Peter C. T. Chen	
PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
US Army ARDEC	
Benet Weapons Laboratory, SMCAR-CCB-TL	AMCMS No. 6111.02.H600.0 PRON No. 1A6DZ602NMSC
Watervliet, NY 12189-4050	
1. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE
US Army ARDEC	July 1987
Close Combat Armaments Center Picatinny Arsenal, NJ 07806-5000	21
4. MONITORING AGENCY NAME & ADDRESS(if different from Controlling Of	fice) 15. SECURITY CLASS. (of this report)
	UNCLASSIFIED
Approved for public release; distribution unli	15. DECLASSIFICATION/DOWNGRADING SCHEDULE
6. DISTRIBUTION STATEMENT (of this Report)  Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the electroct entered in Block 20, if different	mited.
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the ebetract entered in Block 20, 11 difference on Supplementary notes  Presented at the Fourth Army Conference on App	mited.  mi from Report)  lied Mathematics and Computing,
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the electron entered in Block 20, if difference  8. SUPPLEMENTARY NOTES  Presented at the Fourth Army Conference on App Cornell University, Ithaca, New York, 27-30 Ma Published in Proceedings of the Conference.	mited.  Int from Report)  lied Mathematics and Computing, y 1986.
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the electroct entered in Block 20, if difference on Supplementary Notes  Presented at the Fourth Army Conference on App Cornell University, Ithaca, New York, 27-30 Ma Published in Proceedings of the Conference.	mited.  Int from Report)  lied Mathematics and Computing, y 1986.
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the ebetract entered in Block 20, if difference  8. SUPPLEMENTARY NOTES  Presented at the Fourth Army Conference on App Cornell University, Ithaca, New York, 27-30 Ma Published in Proceedings of the Conference.  9. KEY WORDS (Continue on reverse side if necessary and identify by block many plastic Deformation, Tresca's Yield Condition, Flow Rule,	mited.  Int from Report)  lied Mathematics and Computing, y 1986.
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the ebetract entered in Block 20, if difference  8. SUPPLEMENTARY NOTES  Presented at the Fourth Army Conference on Apple Cornell University, Ithaca, New York, 27-30 Ma  Published in Proceedings of the Conference.  9. KEY WORDS (Continue on reverse side if necessary and identity by block in Shrink Fits  Plastic Deformation,  Tresca's Yield Condition,  Flow Rule  Plane Stress	mited.  Int from Report)  lied Mathematics and Computing, y 1986.
Approved for public release; distribution unli  7. DISTRIBUTION STATEMENT (of the ebetract entered in Block 20, if difference  8. SUPPLEMENTARY NOTES  Presented at the Fourth Army Conference on App Cornell University, Ithaca, New York, 27-30 Ma Published in Proceedings of the Conference.  9. KEY WORDS (Continue on reverse side if necessary and identify by block many plastic Deformation, Tresca's Yield Condition, Flow Rule,	mited.  mited.  mited.  lied Mathematics and Computing, y 1986.  miter)  ed herein using an elastic- of different materials. eformations in the ring are con- linear strain-hardening material sociated flow rule. The explicit

UNCLASSIFIED

#### 20. ABSTRACT (CONT'D)

obtained. Numerical results are presented for shrink fit assemblies with different geometric ratio, hardening parameter, and different combinations of materials.

	DTI	
•	BOPY	1
/ INS	SPECT	ED A
	6	

Accesion For	
NTIS CRA&I	A)
DTIC TAB	
Unannon ided	
Justin Cuthan	
, By	
Di. t (5, tio) /	
//www.fubrlity	y Codes
Avdit it	ad for
Dut Spe	Call
A-1	
7 7	

**UNCLASSIFIED** 

WW. Wellekkel (Perkelkkel) pessesse (Persesse) (Persesses Versesses (Persesses (Persesses (Persesses (Persesses

#### TABLE OF CONTENTS

		Page
INTR	ODUCTION	1
ELAS	TIC ASSEMBLY	1
PART	IALLY PLASTIC ASSEMBLY	3
FULL	Y PLASTIC ASSEMBLY	5
NUME	RICAL RESULTS AND DISCUSSIONS	5
REFE	RENCES	8
	LIST OF ILLUSTRATIONS	
1.	Shrink fit assembly.	9
2.	Interference pressure vs. interference for three shrink fit assemblies ( $\alpha$ = 0.5, m = 0.0).	10
3.	Hoop stress at the bore vs. interference for three shrink fit assemblies ( $\alpha$ = 0.5, m = 0.0).	11
4.	The effect of hardening on interference pressure in a composite assembly ( $\alpha$ = 0.5).	12
5.	The effect of hardening on the hoop stress at the bore of a steel ring ( $\alpha$ = 0.5).	13
6.	The effect of geometric ratio on interference pressure in a composite assembly ( $m = 0.05$ ).	14
7.	The effect of geometric ratio on the hoop stress at the bore of a steel ring $(m = 0.05)$ .	15
8.	The distribution of hoop stress in a composite assembly $(\alpha = 0.5, m = 0.0)$ .	16
9.	The distribution of hoop stress in a composite assembly $(\alpha = 0.5, m = 0.1)$ .	17
10.	The distribution of hoop stress in a composite assembly $(\alpha = 0.5, m = 0.2)$ .	18
11.	The distribution of hoop stress in a composite assembly $(\alpha = 1/3, m = 0.1)$ .	19
12.	The distribution of hoop stress in a composite assembly $(\alpha = 0.25, m = 0.1)$ .	20

Occupations and the contraction of the contract of the contrac

#### INTRODUCTION

The shrink fit fastening process is widely used in industry to produce tight, precision assemblies where other fastening methods are neither necessary nor practical. By shrinking a thin ring onto a disk of the same thickness, an elastic state of biaxial, hydrostatic stress can be induced in the disk. For sufficiently small values of interference of the fit, the ring and disk remain elastic; for large values of interference, the ring becomes plastic, first at the interference; for yet larger values of interference, it is possible to produce a plastic state in the disk. This problem was analyzed recently by Gamer and Lance (ref 1) considering the same materials for the disk and ring.

In this report we examine a thin composite shrink fit assembly using a plane-stress elastic-plastic analysis. The ring and disk are made of different materials. Interferences large enough to induce plastic deformations in the ring are considered. The ring material is assumed to be a linear strain-hardening material that obeys Tresca's yield condition and the associated flow rule. The stresses and deformations in the shrink fit assembly are obtained as functions of the interference of the fit.

#### **ELASTIC ASSEMBLY**

A shrink fit assembly is shown in Figure 1. The assembly may be produced by cooling the disk and/or heating the ring with the manufactured interference I. The common interference radius of the assembly is a. The thickness, h, is small compared to a, and hence, the state of stress may be assumed to be plane. All thermal effects are neglected and the displacement is assumed to be small everywhere.

Gamer, U. and Lance, R. H., "Residual Stresses in Shrink Fits," Int. J. Mech. Sci., Vol. 25, No. 7, 1983, pp. 465-470.

For small values of interference of fit, the stress state in the entire assembly is elastic. The stresses and displacements in the ring are

$$\sigma_{\Gamma}$$
 p at at (1a)

$$\frac{\sigma_{\Gamma}}{\pi} = \frac{\rho}{1 - a^{2}/b^{2}} = \frac{a^{2}}{b^{2}} = \frac{a^{2}}{r^{2}}$$
(1a)

$$u/r = (P/E)[(1+\nu)(a^2/r^2) + (1-\nu)(a^2/b^2)]/(1-a^2/b^2)$$
 (1c)

and in the disk

$$\sigma_{\Gamma} = \sigma_{\theta} = -P$$
 ,  $u/r = -(1-\nu_1)P/E_1$  (2)

where E,  $\nu$  and E<sub>1</sub>,  $\nu_1$  are the material constants of the ring and disk, respectively. At the interface,  $u_a$  (ring) -  $u_a$  (disk) = I by the compatibility requirement. The interference pressure (p) is a function of the interference (I) given by

$$p = \frac{EI}{a} \left(1 - \frac{a^2}{b^2}\right) / \left[ (1+\nu) + (1-\nu) \frac{a^2}{b^2} + (1-\nu_1) \left(1 - \frac{a^2}{b^2} + \frac{E}{E_1} \right) \right]$$
(3)

For sufficiently large values of the interference, the stresses in the ring reach the yield limit. Assuming that Tresca's yield condition governs the behavior of the material, the ring first becomes plastic at the interference when the stresses satisfy

$$\sigma_{\theta} - \sigma_{\Gamma} = \sigma_{0} \tag{4}$$

where  $\sigma_0$  is the initial tensile yield stress. The solution for the critical interference pressure to cause incipient plastic deformation is

$$p^* = \frac{1}{2} \sigma_0 (1 - a^2/b^2) \tag{5}$$

and it follows from Eq. (3) that the interference for the onset of plastic flow is

$$I^* = \frac{\sigma_0}{E} \frac{a}{2} \left[ (1+\nu) + (1-\nu) \frac{a^2}{b^2} + (1-\nu_1)(1 - \frac{a^2}{b^2}) \frac{E}{E_1} \right]$$
 (6)

which reduces to I\* =  $a\sigma_0/E$  for the special case (E<sub>1</sub> = E,  $\nu_1$  =  $\nu$ ) considered in Reference 1.

#### PARTIALLY PLASTIC ASSEMBLY

For values of interference larger than that given by Eq. (4), a plastic zone forms in the ring, so that for a  $\leq r \leq \rho$  the ring is plastic, while for  $\rho \leq r \leq \rho$ , the ring material is still in an elastic state. The elastic-plastic interface radius  $\rho$  is a function of the interference I.

We assume that the ring is made of a linear work-hardening material which obeys Tresca's yield condition

$$\sigma_{\theta} - \sigma_{r} = \sigma \tag{7}$$

where the yield stress  $\sigma$  is a function of the plastic strain  $\epsilon^p$ . For a linear work-hardening material, we have

$$\sigma = \sigma_0 (1 + \eta \epsilon^p)$$
 and  $\eta = (E/\sigma_0)m/(1-m)$  (8)

where  $\eta$  (or m) is the hardening parameter.

Applying the usual flow rule and following the method of analysis reported by Gamer and Lance (ref 1) and Bland (ref 2), the expressions for the stresses and the displacement can be obtained explicitly. The complete solution in a  $\leqslant p$  is:

$$\sigma_{\Gamma} = \sigma_{O}(1-m)[2n - \frac{r}{a} - \frac{1}{2}(1-\nu)\eta - \frac{D}{E}r^{-2}] + C$$
 (9)

$$\sigma_{\theta} = \sigma_{0}(1-m)[1+2n - + \frac{r}{2} + \frac{r}{2}(1-\nu)\eta - \frac{r}{r} + \frac{r}{2}] + C \qquad (10)$$

$$(1-\nu)^{-1}u = \frac{\sigma_0}{E} (1-m) [r!n - \frac{\nu}{a} - \frac{\nu}{4}(1-\nu)\eta - \frac{\nu}{E} - \frac{\nu}{E}] + \frac{\nu}{E} + \frac{\nu}{E} - \frac{\nu}{E}$$
(11)

Gamer, U. and Lance, R. H., "Residual Stresses in Shrink Fits," Int. J. Mech. Sci., Vol. 25, No. 7, 1983, pp. 465-470.

<sup>&</sup>lt;sup>2</sup>Bland, D. R., "Elastoplastic Thick-Walled Tubes of Work-Hardening Materials Subject to Internal and External Pressures and Temperature Gradients," <u>J. Mech. Phys. Solids</u>, Vol. 4, 1956, pp. 209-229.

In the elastic zone,  $\rho \le r \le b$ , the stresses and the displacement are:

$$r$$
  $E$   $E$   $B$   $(12)$ 

$$\frac{\sigma_{\Gamma}}{\sigma_{\theta}} = \frac{E}{1-\nu} A \mp \frac{E}{1+\nu} B \qquad (12)$$

$$\frac{\sigma_{\theta}}{\sigma_{\theta}} = \frac{1-\nu}{1+\nu} A \mp \frac{E}{1+\nu} B \qquad (13)$$

$$u = Ar + B/r \tag{14}$$

The constants A, B, C, D, p, and  $\rho$  all depend on the interference I, and can be evaluated by considering the following conditions: continuity of stress and displacement at  $r = \rho$  requires  $\sigma_r(\rho^-) = \sigma_r(\rho^+)$  and  $u(\rho^-) = u(\rho^+)$ . At the ring-disk interface  $\sigma_r(a) = -p$  and at the outer surface of the ring  $\sigma_r(b) = 0$ . The yield condition in Eq. (7) must be satisfied at  $r = \rho$  and finally, compatibility of the displacement field with the interference I requires that u(a+) -  $u(a^-) = I$ . These conditions are sufficient to determine all unknown parameters. In this report the constants A, B, C, D are determined as functions of ho.

$$A = \frac{1}{2}(1-\nu)(\sigma_0/E)(\rho/b)^2$$
,  $B = \frac{1}{2}(1+\nu)(\sigma_0/E)\rho^2$ 

$$C = \sigma_0[\frac{1}{2}m - (1-m)\ln(b/a) - \frac{1}{2}(1-\rho^2/b^2)] , D = \sigma_0\rho^2/(1-\nu)$$
 (15)

The dimensionless interference pressure and interference are given, respectively, by

$$\bar{p} = P/\sigma_0 = \frac{1}{2}(1-\rho^2/b^2) + (1-m)\ln(\rho/a) + \frac{1}{2}m(\rho^2/a^2-1)$$
 (16)

$$\bar{I} = (E/\sigma_0)I/a = (\rho/a)^2 - [(1-\nu) - (1-\nu_1)E/E_1](P/\sigma_0)$$
 (17)

When the ring and disk are made of the same material, i.e.,  $E_1$  = E,  $v_1$  = v, Eq. (17) reduces to the simple formula,  $(E/\sigma_0)I/a = (\rho/a)^2$ . For this special case (ref 1), the constants A, B, C, D, P, and  $\rho$  can be expressed explicitly as functions of interference I. In general, the interference pressure (p) is related to the interference (I) implicitly through the elastic-plastic interface  $(\rho)$  as shown in Eqs. (16) and (17) for a  $\leq \rho \leq b$ . The upper limit of the partially

<sup>&</sup>lt;sup>1</sup>Gamer, U. and Lance, R. H., "Residual Stresses in Shrink Fits," <u>Int. J. Mech.</u> <u>Sci.</u>, Vol. 25, No. 7, 1983, pp. 465-470.

plastic assembly is obtained by letting  $\rho = b$ . The corresponding interference pressure (p\*\*) and interference (I\*\*) are

$$p^{**}/\sigma_{O} = (1-m) \ln(b/a) + \frac{1}{4} m (b^{2}/a^{2}-1)$$

$$(E/\sigma_{O}) I^{**}/a = (b/a)^{2} - [(1-\nu) - (1-\nu_{1}) E/E_{1}] p^{**}/\sigma_{O}$$
(18)

#### FULLY PLASTIC ASSEMBLY

When the interference I is larger than I\*\*, we have reached the fully plastic state in the ring. In this case, the expressions for the stresses and the displacement in a  $\leq r \leq b$  are still the same as those given by Eqs. (9), (10), and (11). The constants C, D, and the interference p are determined with the boundary conditions  $\sigma_{\Gamma}(a) = -p$ ,  $\sigma_{\Gamma}(b) = 0$ , and the compatibility requirement  $u(a^+) - u(a^-) = I$ . The results for the constants are

$$C = [pa^{2}/b^{2} - (1-m)\sigma_{0} \ln(b/a)]/(1-a^{2}/b^{2})$$

$$D = 2a^{2}[p - (1-m)\sigma_{0}\ln(b/a)]/[m(1-\nu)(1-a^{2}/b^{2})]$$
(19)

and the interference pressure is given as a function of interference by

AND SOME THE STATE OF THE SECREPT OF

#### NUMERICAL RESULTS AND DISCUSSIONS

The analysis described above makes it possible to predict the interference pressure in a composite shrink fit assembly, and hence, determine the stress state in the ring and disk as a function of the interference. The numerical results have been obtained for shrink fit assemblies with different geometric ratio ( $\alpha$  = a/b), hardening parameter (m), and different combinations of materials. For a steel ring with  $\alpha$  = 0.5, m = 0.0, E = 30x10° psi,  $\nu$  = 0.3,  $\sigma_0$  = 15x10° psi, we have considered three types of disks: (a) rigid disk with E<sub>1</sub> = 1000 E,  $\nu_1$  = 0.0,  $\sigma_1$  = 1000  $\sigma_0$ ; (b) steel disk of the same material as the ring;

(c) a disk made of tungsten carbide with  $E_1$  = 88.5x10° psi,  $\nu_1$  = 0.258,  $\sigma_1$  = 50x10° psi. The numerical results of the interference pressure (p/ $\sigma_0$ ) for these three cases are presented graphically in Figure 2 as functions of the interference ( $\overline{I}$ ). The results of the hoop stress at the inside surface of the ring are presented in Figure 3 also for these three cases. As can be seen from these two figures, the results for the composite shrink fit assembly marked (c) fall between the two limits established by cases (a) and (b).

For composite shrink fit assemblies made of tungsten carbide disk and steel ring with  $\alpha=0.5$ , m=0.0, 0.1, 0.2, the results are presented in Figures 4 and 5, respectively, for the interference pressure and the hoop stress at the bore as functions of the interference. The effect of hardening parameter (m) on these relations can be seen from these two figures. For the same combination of composite shrink fit assembly with m=0.05,  $\alpha=1/4$ , 1/3, 1/2, 3/4, the results showing the effect of geometric ratio ( $\alpha$ ) are shown in Figures 6 and 7 for the interference pressure and hoop stress at the bore, respectively.

The numerical results of the stresses and displacements in composite shrink fit assemblies have also been obtained, but only some results are presented here. The distributions of hoop stresses in a steel ring with  $\alpha=0.5$  are shown in Figures 8, 9, and 10 for m = 0.0, 0.1, 0.2, respectively. In each figure, we have shown the results corresponding to four stages of interference: (a) initial yielding ( $\rho/a=1.0$ ),  $\bar{I}^*=0.832$ ; (b) partial yielding ( $\rho/a=1.5$ ),  $\bar{I}=1.970$ , 1.960, 1.950; (c) complete yielding ( $\rho/a=2.0$ ),  $\bar{I}^**=3.689$ , 3.653, 3.617; (d) fully plastic state with  $\bar{I}=1.5$  I\*\*. For an ideally plastic ring (m = 0.0), the stress distribution remains unchanged after complete yielding has been reached. For stra...-hardening rings, the stress distributions show large variations, especially for large values of interference. As shown in Figures 8,

9, and 10, the hardening parameter has a significant effect on the stress distributions. Additional stress distributions in the ring with m=0.1 are shown in Figures 11 and 12 for  $\alpha=1/3$  and 1/4, respectively. The effect of geometric ratio on the distributions can be seen by comparing Figures 9, 11, and 12.

#### REFERENCES

- 1. Gamer, U. and Lance, R. H., "Residual Stresses in Shrink Fits," Int. J. Mech. Sci., Vol. 25, No. 7, 1983, pp. 465-470.
- 2. Bland, D. R., "Elastoplastic Thick-Walled Tubes of Work-Hardening Materials Subject to Internal and External Pressures and Temperature Gradients," J. Mech. Phys. Solids, Vol. 4, 1956, pp. 209-229.

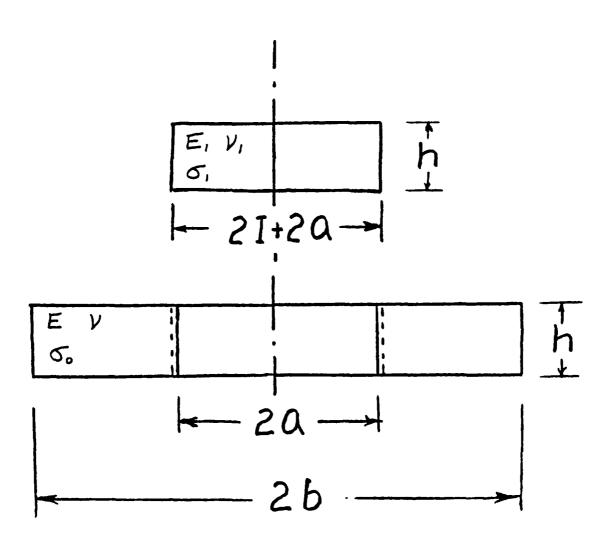
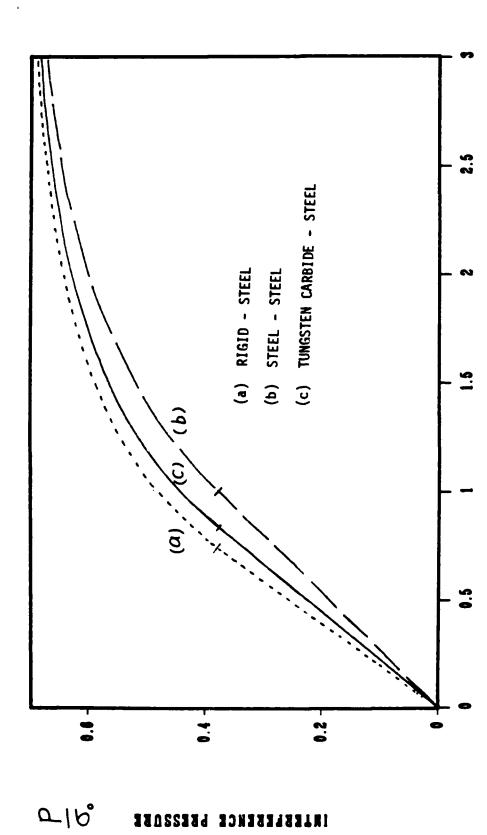


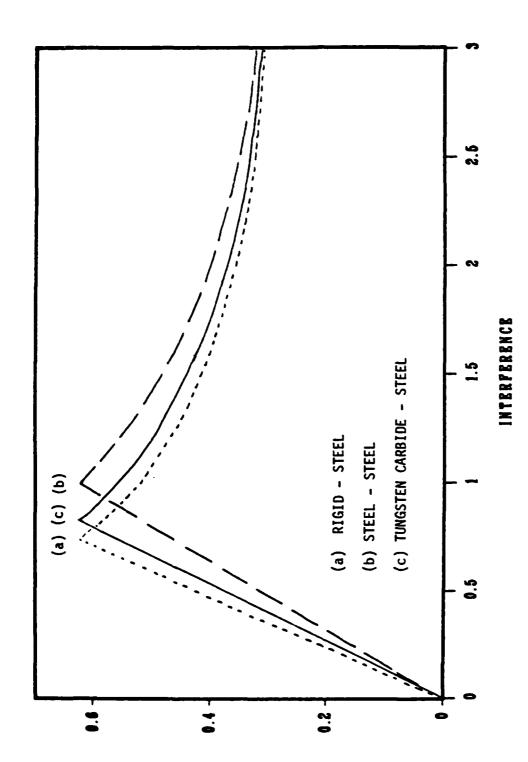
Figure 1. Shrink fit assembly.



Interference pressure vs. interference for three shrink fit assemblies ( $\alpha$  = 0.5, m = 0.0). Figure 2.

E1/5.a

INTERPERENCE

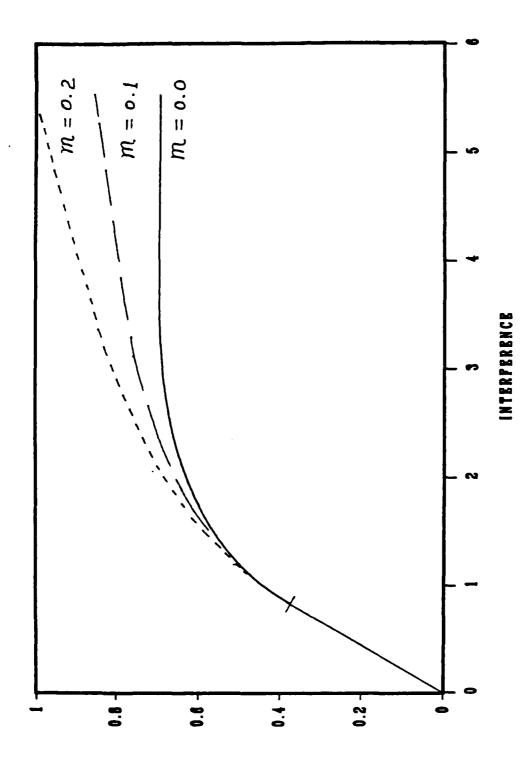


Hoop stress at the bore vs. interference for three shrink fit assemblies ( $\alpha = 0.5$ , m = 0.0). Figure 3.

ESCENDE DESCRIPTION DE L'ESCENTIN DE L'ESCEN

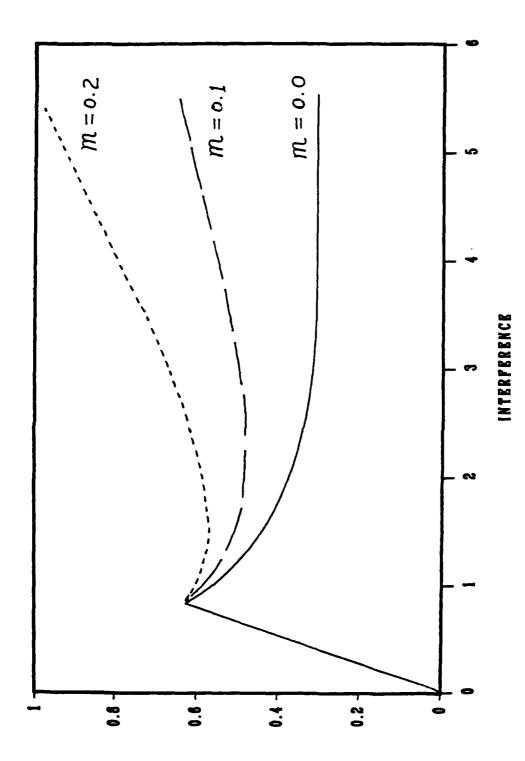
Parameter (1997) Teach

SIEESS VI THE BOEE



The effect of hardening on interference pressure in a composite assembly ( $\alpha$  = 0.5). Figure 4.

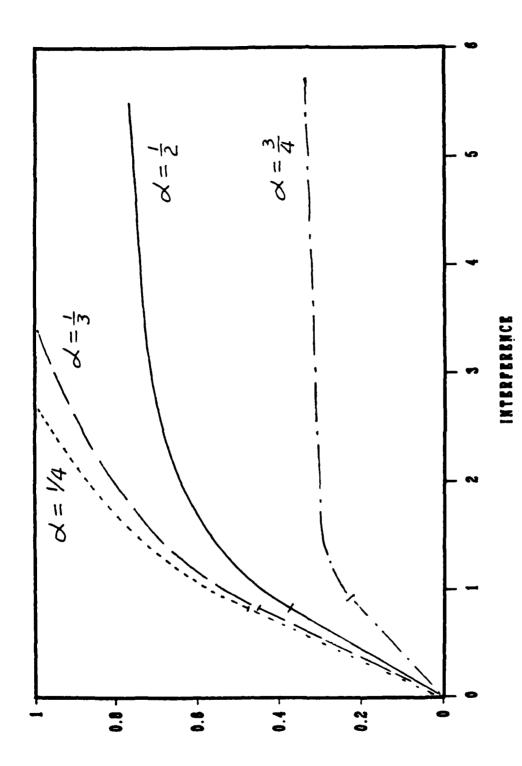
INTERPRENCE PRESSURE



The effect of hardening on the hoop stress at the bore of steel ring  $(\alpha = 0.5)$ . Figure 5.

SONDER SON CONTRACTOR LIVER SON CONTRACTOR DANS SON PROCESSON PROCESSON DESCRIPE PROCESSON PROCESSON DESCRIPE

HOOP STRESS AT THE BORE



The effect of geometric ratio on interference pressure in a composite assembly (m = 0.05). Figure 6.

INTERPRENCE PRESSURE

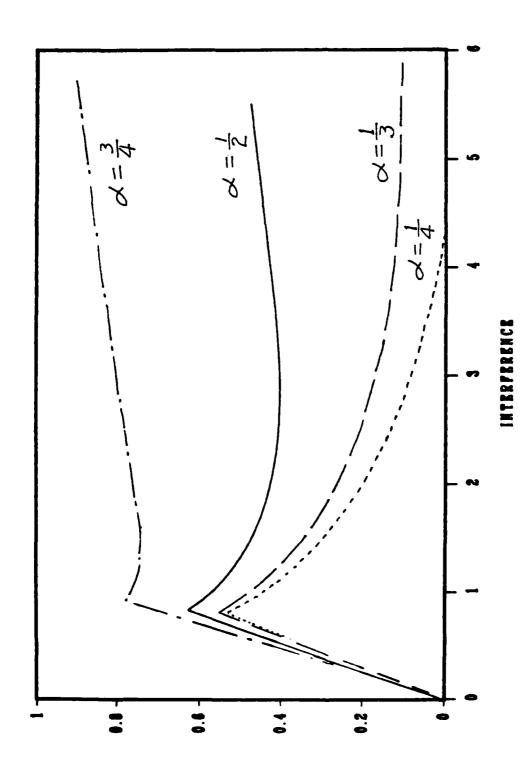


Figure 7. The effect of geometric ratio on the hoop stress at the bore of a steel ring ( $\mathbf{m} = 0.05$ ).

HOOF STRESS AT THE BORE

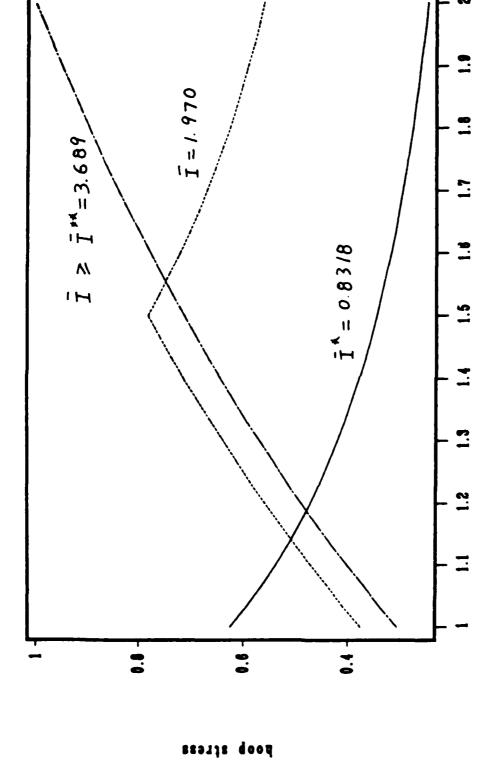


Figure 8. The distribution of hoop stress in a composite assembly  $(\alpha = 0.5, \, \text{m} = 0.0).$ 

radial distance

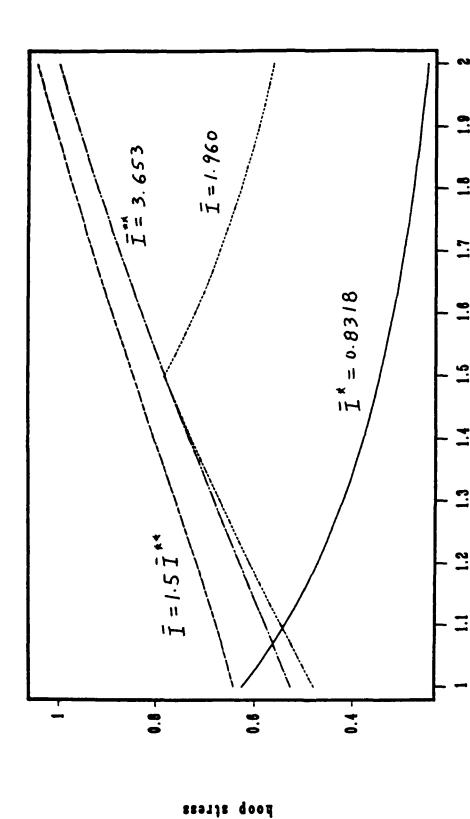


Figure 9. The distribution of hoop stress in a composite assembly  $\{\alpha=0.5,\ m=0.1\}.$ 

radial distance

A STATE POSSO DE POSS

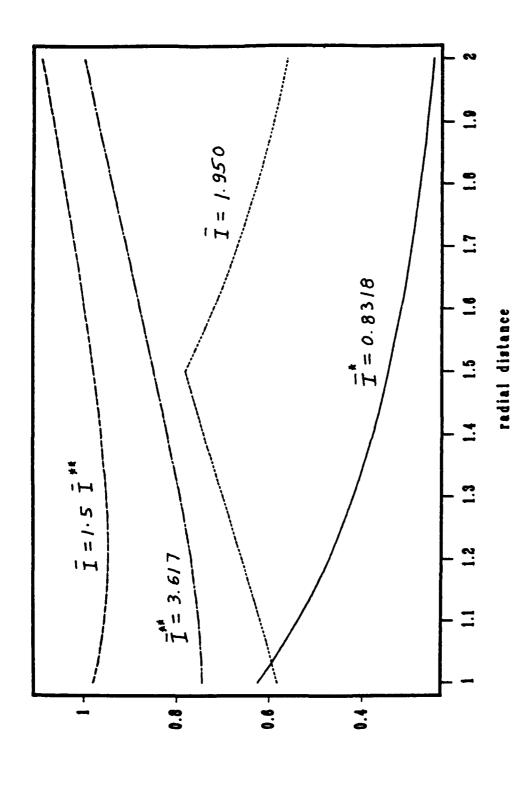


Figure 10. The distribution of hoop stress in a composite assembly (  $\alpha$  = 0.5, m = 0.2).

Foob stress

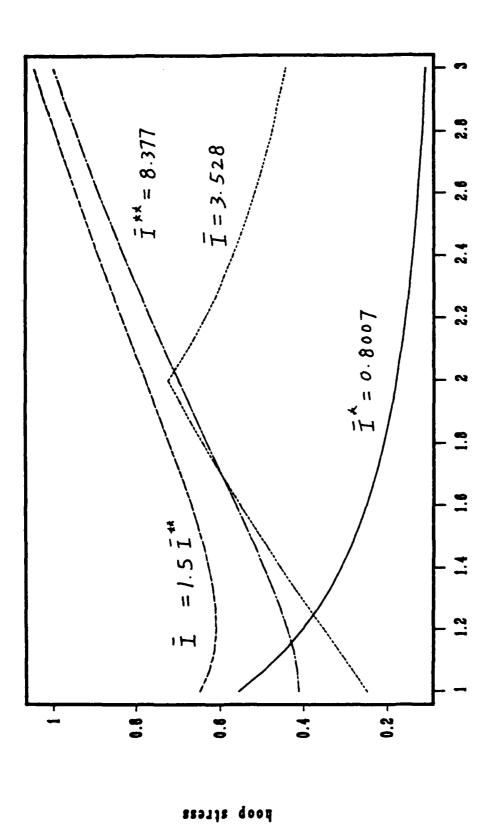


Figure 11. The distribution of hoop stress in a composite assembly  $(\alpha\,=\,1/3,\,\,m\,=\,0.1).$ 

radial distance

CONTROL CONTROL OF SECURIOR PROPERTY OF SECURIOR PR

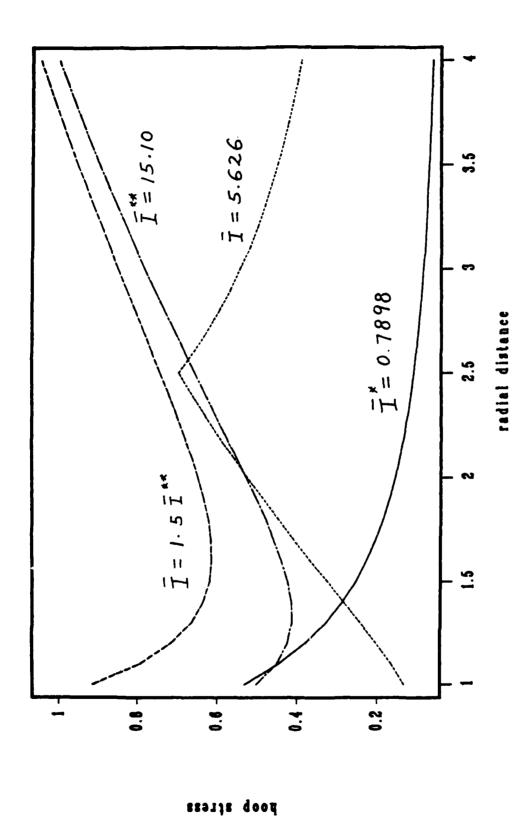


Figure 12. The distribution of hoop stress in a composite assembly  $(\alpha = 0.25, \ \text{m} = 0.1).$ 

#### TECHNICAL REPORT INTERNAL DISTRIBUTION LIST

	NO. OF COPIES
CHIEF, DEVELOPMENT ENGINEERING BRANCH	
ATTN: SMCAR-CCB-D	1
-DA	1
-DC	1
-DM	1
-DP	1
-DR	1
-DS (SYSTEMS)	1
CHIEF, ENGINEERING SUPPORT BRANCH	
ATTN: SMCAR-CCB-S	1
-SE	1
CHIEF, RESEARCH BRANCH	
ATTN: SMCAR-CCB-R	2
-R (ELLEN FOGARTY)	1
-RA	1 1
-RM	1
-RP	1
-RT	1
TECHNICAL LIBRARY	5
ATTN: SMCAR-CCB-TL	
TECHNICAL PUBLICATIONS & EDITING UNIT	2
ATTN: SMCAR-CCB-TL	
DIRECTOR, OPERATIONS DIRECTORATE	1
ATTN: SMCWV-OD	
DIRECTOR, PROCUREMENT DIRECTORATE	1
ATTN: SMCWV-PP	
DIRECTOR, PRODUCT ASSURANCE DIRECTORATE	1
ATTN: SMCWV-QA	

NOTE: PLEASE NOTIFY DIRECTOR, BENET WEAPONS LABORATORY, ATTN: SMCAR-CCB-TL, OF ANY ADDRESS CHANGES.

#### TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST

	NO. OF COPIES		NO. OF COPIES
ASST SEC OF THE ARMY RESEARCH AND DEVELOPMENT ATTN: DEPT FOR SCI AND TECH THE PENTAGON WASHINGTON, D.C. 20310-0103	1	COMMANDER ROCK ISLAND ARSENAL ATTN: SMCRI-ENM ROCK ISLAND, IL 61299-5000	1
ADMINISTRATOR DEFENSE TECHNICAL INFO CENTER ATTN: DTIC-FDAC CAMERON STATION	12	DIRECTOR US ARMY INDUSTRIAL BASE ENGR ACT ATTN: AMXIB-P ROCK ISLAND, IL 61299-7260	rv I
ALEXANDRIA, VA 22304-6145  COMMANDER US ARMY ARDEC ATTN: SMCAR-AEE	1	COMMANDER US ARMY TANK-AUTMV R&D COMMAND ATTN: AMSTA-DDL (TECH LIB) WARREN, MI 48397-5000	1
SMCAR-AES, BLDG. 321 SMCAR-AET-O, BLDG. 351N SMCAR-CC SMCAR-CCP-A	1 1 1	COMMANDER US MILITARY ACADEMY ATTN: DEPARTMENT OF MECHANICS WEST POINT, NY 10996-1792	1
SMCAR-FSA SMCAR-FSM-E SMCAR-FSS-D, BLDG. 94 SMCAR-MSI (STINFO) PICATINNY ARSENAL, NJ 07806-5000	1 1 1 2	US ARMY MISSILE COMMAND REDSTONE SCIENTIFIC INFO CTR ATTN: DOCUMENTS SECT, BLDG. 4484 REDSTONE ARSENAL, AL 35898-5241	2
DIRECTOR US ARMY BALLISTIC RESEARCH LABORAT ATTN: SLCBR-DD-T, BLDG. 305 ABERDEEN PROVING GROUND, MD 21005- DIRECTOR	1	COMMANDER US ARMY FGN SCIENCE AND TECH CTR ATTN: DRXST-SD 220 7TH STREET, N.E. CHARLOTTESVILLE, VA 22901	1
US ARMY MATERIEL SYSTEMS ANALYSIS ATTN: AMXSY-MP ABERDEEN PROVING GROUND, MD 21005- COMMANDER HQ, AMCCOM ATTN: AMSMC-IMP-L	1	COMMANDER US ARMY LABCOM MATERIALS TECHNOLOGY LAB ATTN: SLCMT-IML (TECH LIB) WATERTOWN, MA 02172-0001	2
ROCK ISLAND, IL 61299-6000			

NOTE: PLEASE NOTIFY COMMANDER, ARMAMENT RESEARCH, DEVELOPMENT, AND ENGINEERING CENTER, US ARMY AMCCOM, ATTN: BENET WEAPONS LABORATORY, SMCAR-CCB-TL, WATERVLIET, NY 12189-4050, OF ANY ADDRESS CHANGES.

#### TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST (CONT'D)

	NO. OF <u>COPIES</u>		NO. OF COPIES
COMMANDER		COMMANDER	
US ARMY LABCOM, ISA		AIR FORCE ARMAMENT LABORATORY	
ATTN: SLCIS-IM-TL	1	ATTN: AFATL/MN	1
2800 POWDER MILL ROAD		EGLIN AFB, FL 32543-5434	
ADELPHI, MD 20783-1145			
		COMMANDER	
COMMANDER		AIR FORCE ARMAMENT LABORATORY	
US ARMY RESEARCH OFFICE		ATTN: AFATL/MNG	
ATTN: CHIEF, IPO	1	EGLIN AFB, FL 32542-5000	1
P.O. BOX 12211			
RESEARCH TRIANGLE PARK, NC	27709-2211	METALS AND CERAMICS INFO CTR	
		BATTELLE COLUMBUS DIVISION	
DIRECTOR		505 KING AVENUE	
US NAVAL RESEARCH LAB		COLUMBUS, OH 43201-2693	1
ATTN: DIR, MECH DIV	1		
CODE 26-27 (DOC LIB)	1		
WASHINGTON, D.C. 20375			

NOTE: PLEASE NOTIFY COMMANDER, ARMAMENT RESEARCH, DEVELOPMENT, AND ENGINEERING CENTER, US ARMY AMCCOM, ATTN: BENET WEAPONS LABORATORY, SMCAR-CCB-TL, WATERVLIET, NY 12189-4050, OF ANY ADDRESS CHANGES.

A